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# Design and analysis of bolted joints

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# Learning Objectives

## Part 1:

- To be familiar with the **types of joints** including joining technologies, loading conditions, and design considerations of bolted joints

## Part 2:

- To be able to determine the **stiffness** of the bolt & clamped members, **resultant loads** of pre-tensioned joint
- To be able to design against **joint failure** & to calculate **reserve factor** of a pre-tensioned joint
- To understand the behaviour of pre-tensioned joint under **cyclic loading**

## Part 3:

- **Worked examples**

# Function & types of Joints/Fasteners

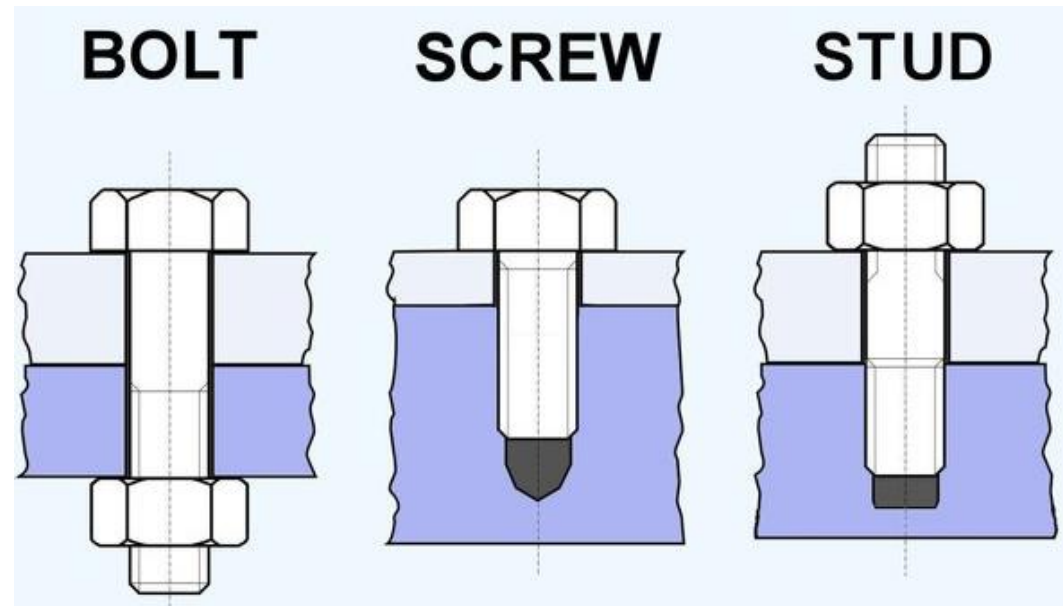
- A **joint** or **fastener** is a device used to connect or join two or more components.

A layman might consider **threaded fasteners** uninteresting machine elements, but it is **unimaginable** that most machines & structures can be built without.

- **Non-permanent joints**
  - Screws, threaded and unthreaded fasteners and setscrews, etc.
- **Permanent joints**
  - Rivets, welded and adhesive joints, friction joints, etc.
  - Fasteners used as permanent joints that are never disassembled.

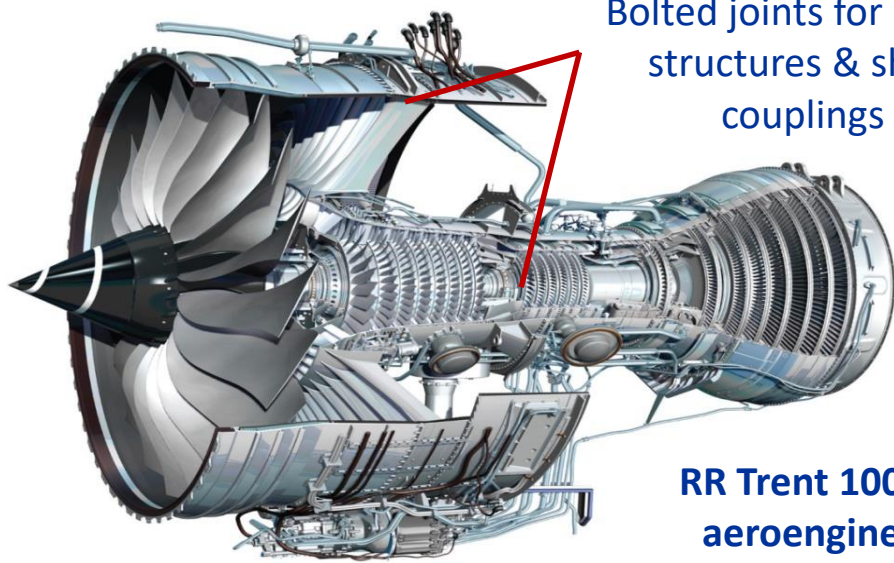
# A bolt, a screw or a stud?

- A **bolt** is a threaded fastener mated with a nut.
- A **screw** has either pre-formed or self-made internal threads.
- A **stud** is an externally threaded headless fastener. One end mates with a tapped component and the other with a standard nut.

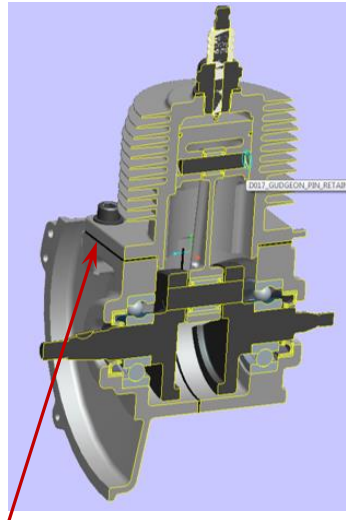


Bolts, screws and studs are all commonly used in different applications, the general design consideration and method are similar.

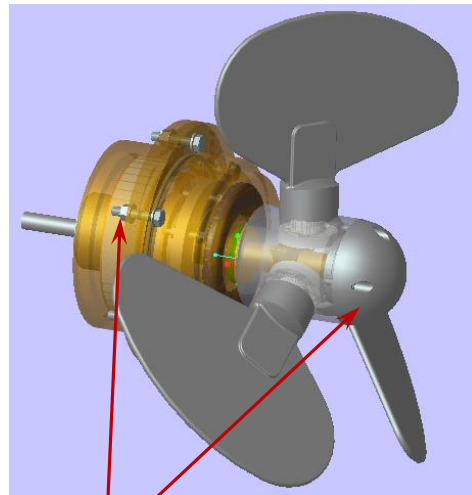
# Examples of bolted joints



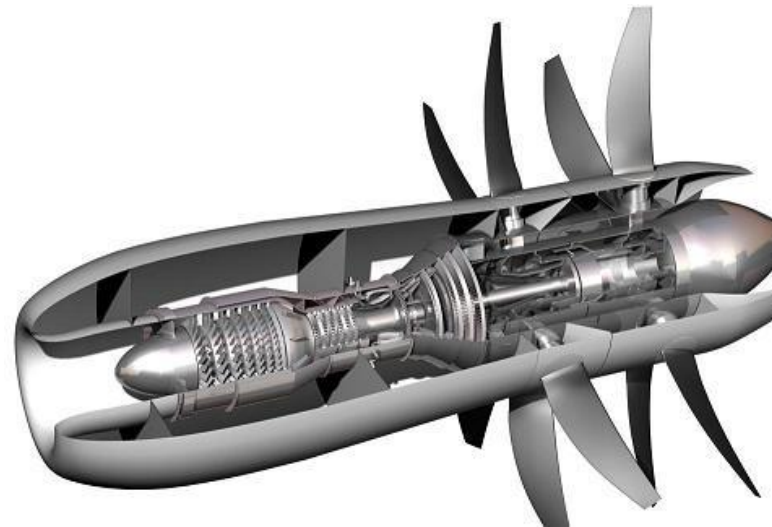
Wind turbine blades joined by bolts to the hub



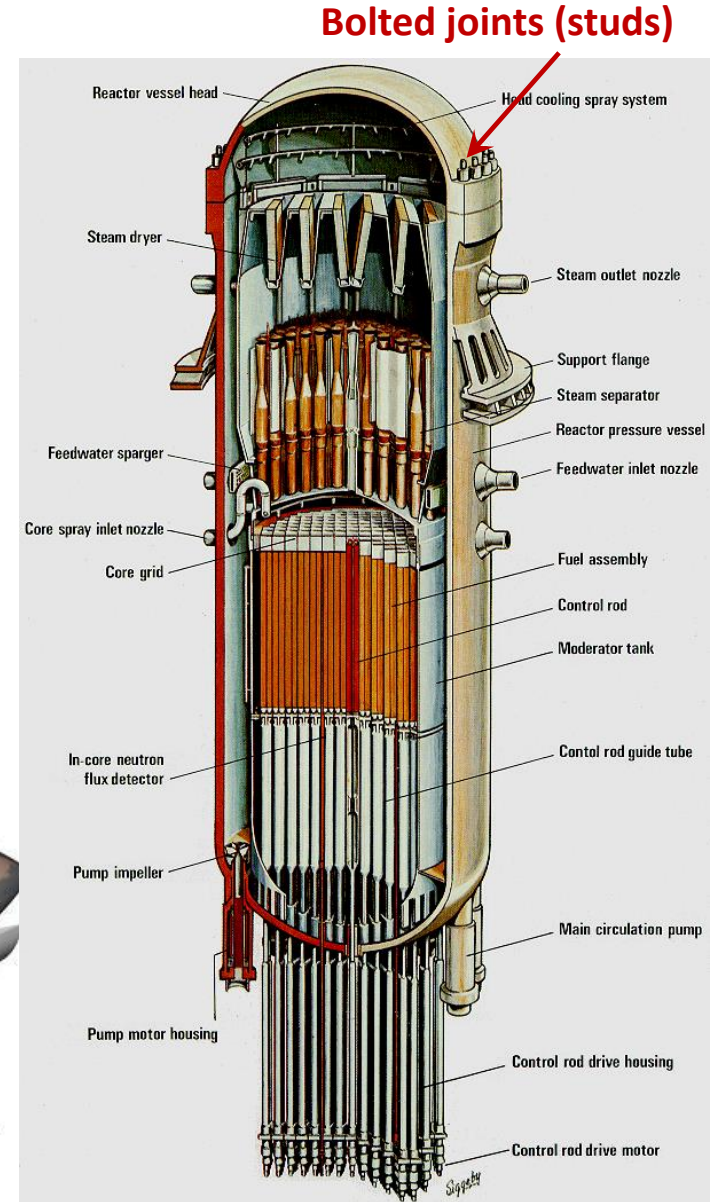
Bolted joints (socket screw) of a two-stroke engine



Bolted joints of a variable pitch propeller



A past research project with RR in design optimisation of Open rotor jet engine

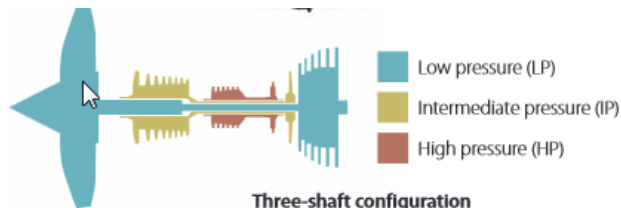
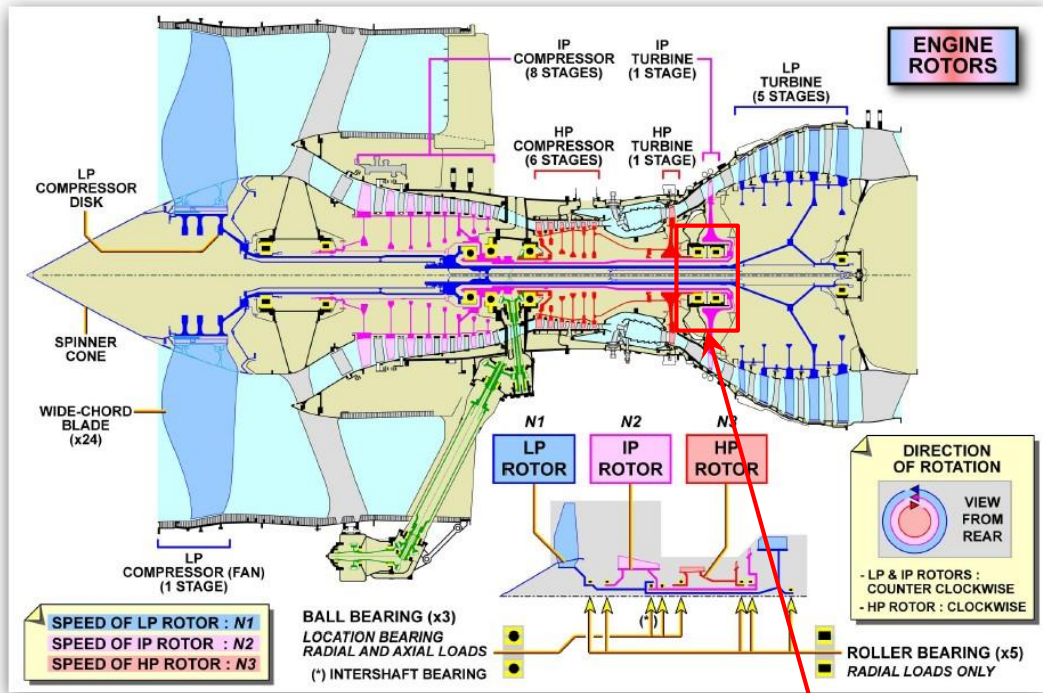


Nuclear Reactor Pressure Vessel

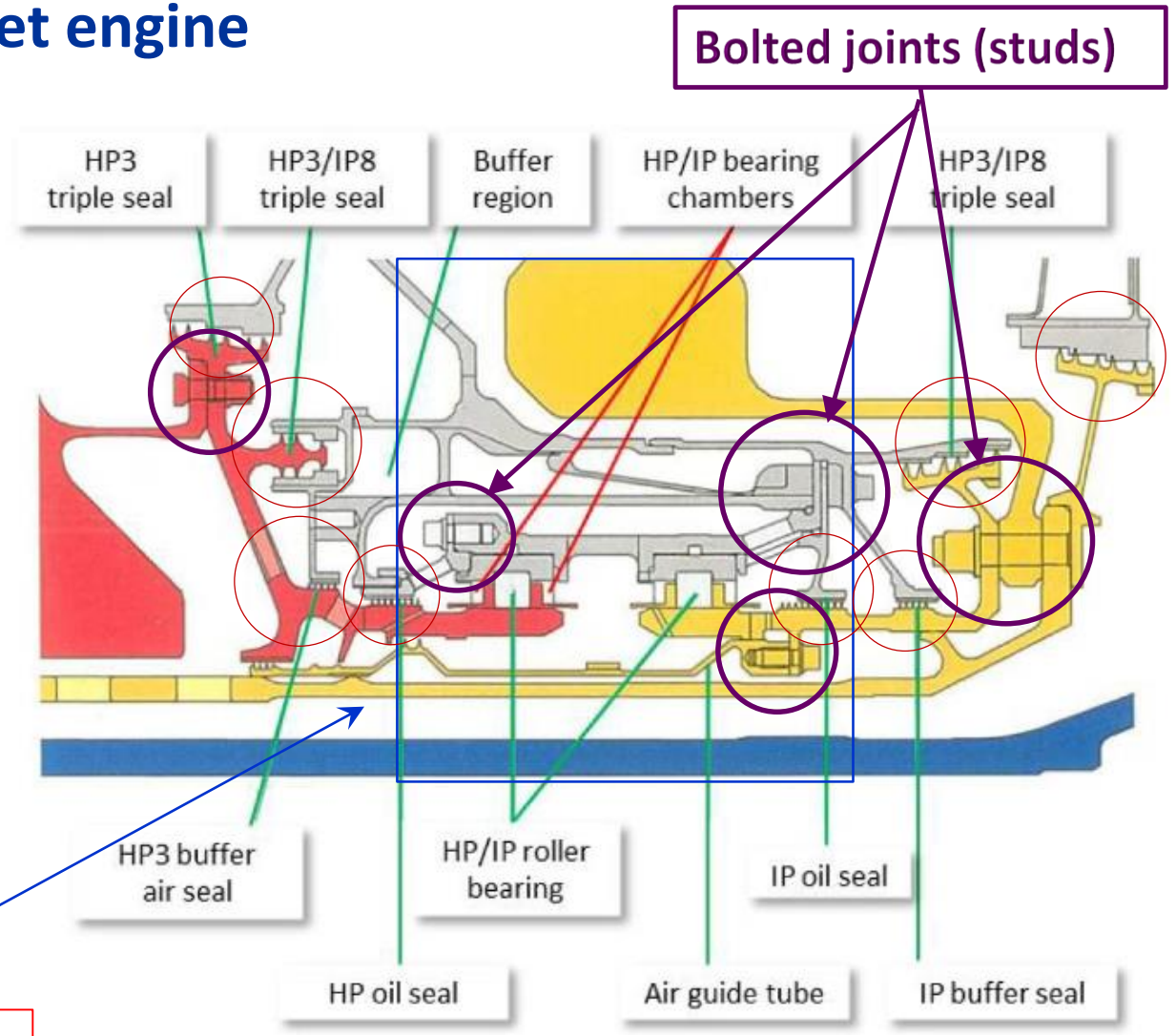


# Labyrinth seals in aeroengine applications

- Rolls-Royce Trent 1000 three shaft jet engine

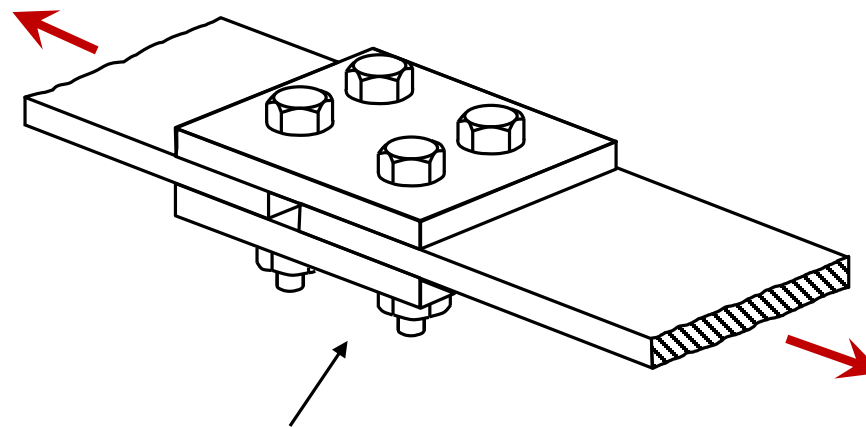
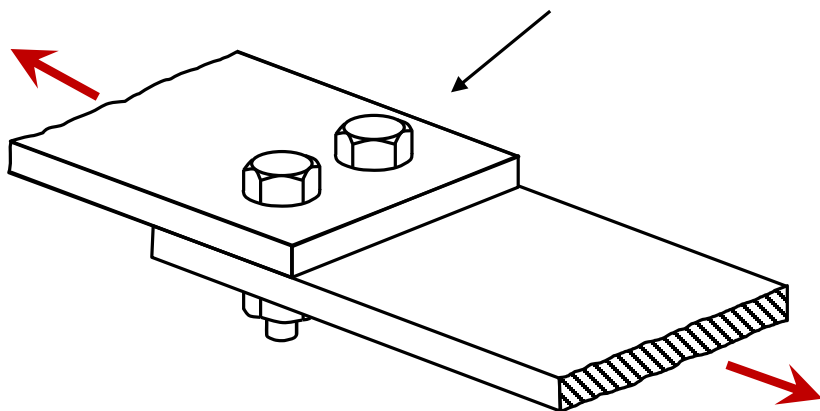


**HP-IP bearing chamber**

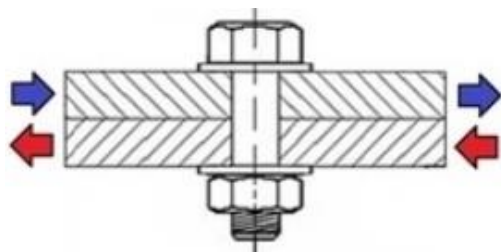


# Shear joints

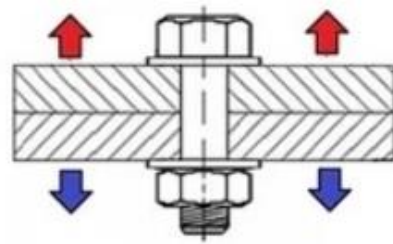
Components to be joined overlap each other in a **lap joint**



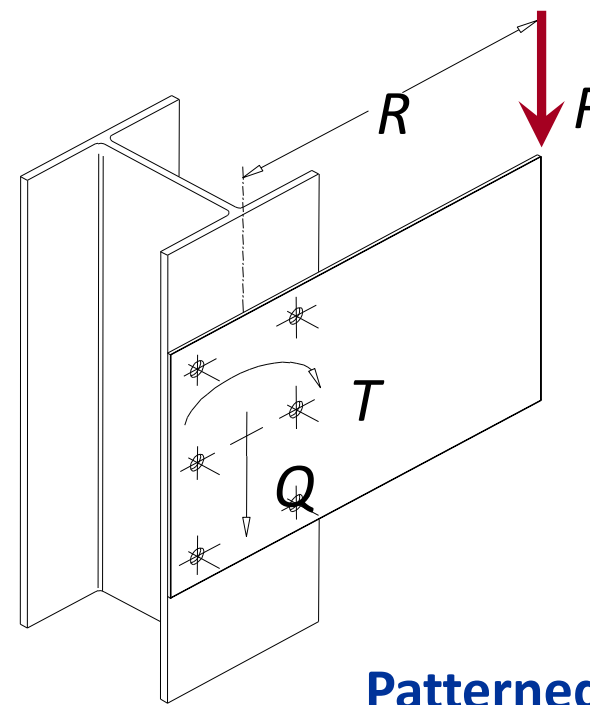
An additional component is used to bridge the gap in a **butt joint**



Shear joints



Bolted joints under axial loading



Patterned joints

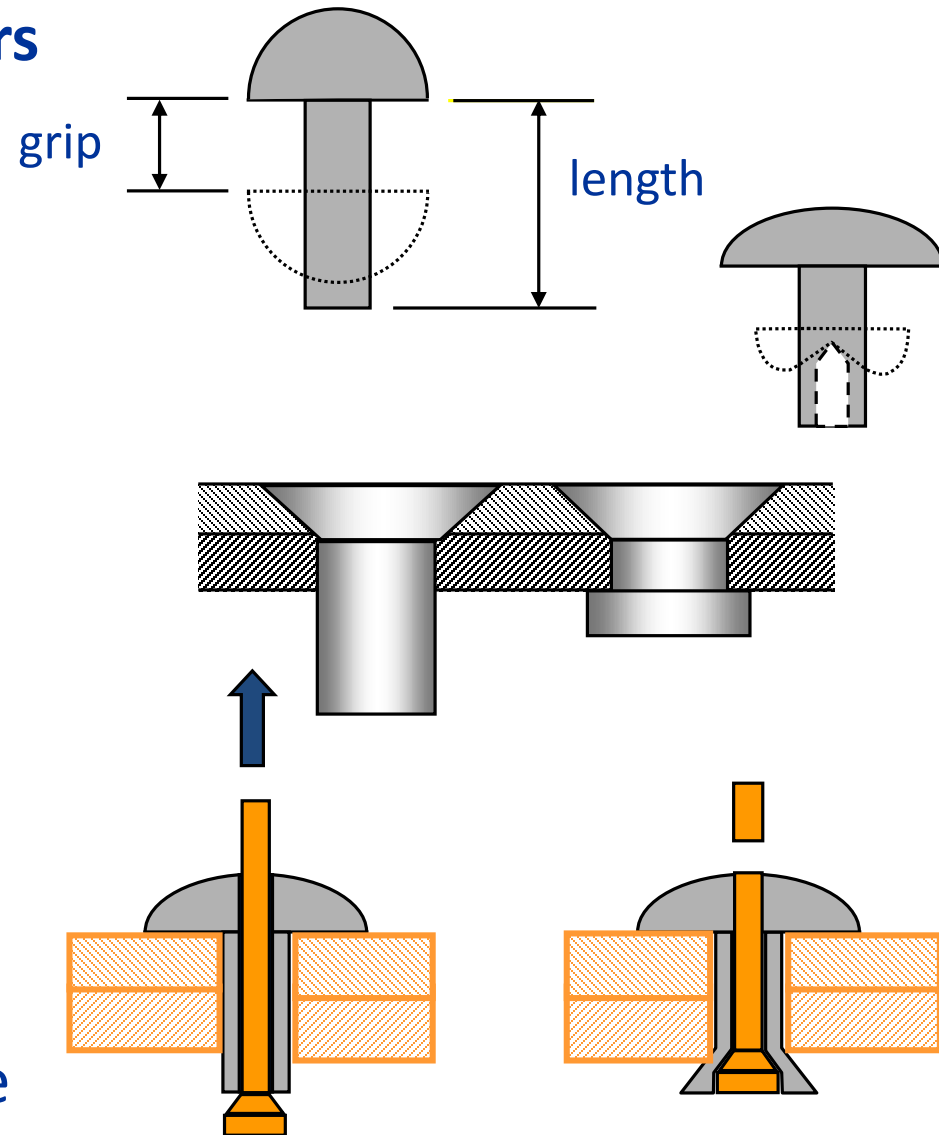
Note: Design and calculation of **shear joints** should use a different approach. It is not covered in this session.

Refer Shigley's Mechanical Engineering Design, Ch8 for detail

# Rivets

## Rivets are non-threaded fasteners

- **Solid rivets** – assembled from one side and ‘upset’ on the other
  - **BS 4620: 1970** for general purposes
  - For airframe applications BS A 361, 351(Ni), 362(Ti).
- **Tubular rivets** have a hole down the axis – upsetting is easier
- **Blind rivets** are tubular and upset from the installation side





# Examples of Riveted structures



- Number of rivets  $\approx$  6,000,000
- Largest rivet is 3.5 kilograms, 395 mm long



**7,500,000 rivets used**

# More example of Rivets

- Rivets remain an important method of assembly of aircraft structures
- More or less permanent, they don't work loose
- But can be drilled out if repairs are needed
- **Boeing 747** uses over  $2.5 \times 10^6$  fasteners



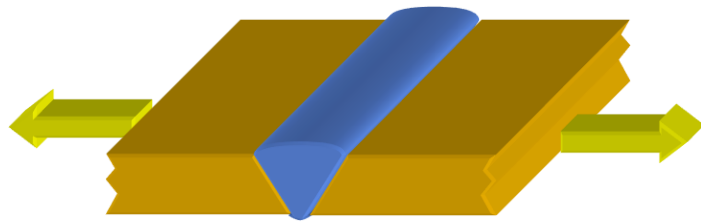
# Rivets

- **Advantages:**
  - Low cost
  - Rapid assembly (thousands/hour when automated)
  - Permanent (but can be drilled out)
  - Can join dissimilar materials (watch out for galvanic corrosion)
  - Wide range of shapes and materials
- **Disadvantages:**
  - Slow compared with welding and adhesives
  - **Shear loading (poor in tensile loading)**
  - Joints leak unless sealed

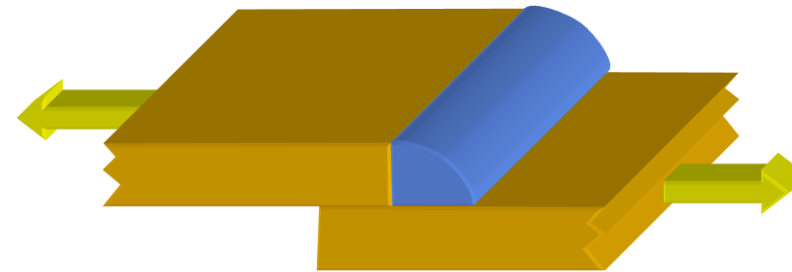
# Welded joints

**Welding** is a fabrication process that joins materials by using **high heat** to melt the parts and filler material together.

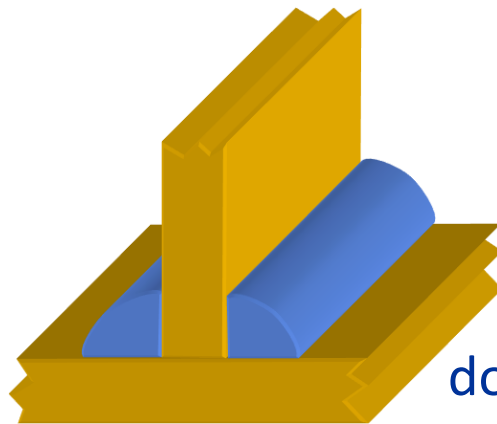
**Gas flame, electric arc, laser, electron beam, friction and ultrasound** can all be used as energy sources for heating.



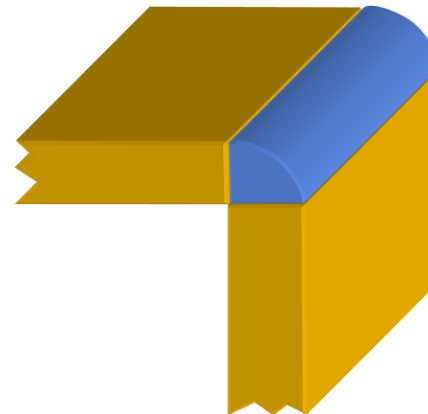
V groove or **butt joint**



single fillet **lap joint**



double fillet  
**T joint**



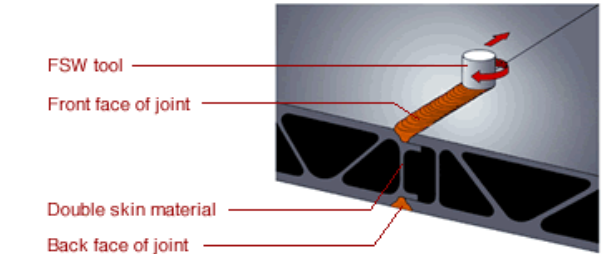
fillet welded  
**corner joint**



# Examples of welded joints



Airbus 380, lower fuselage panels (**laser welding**)



Hitachi High speed train  
(**Friction Stir Welding**)



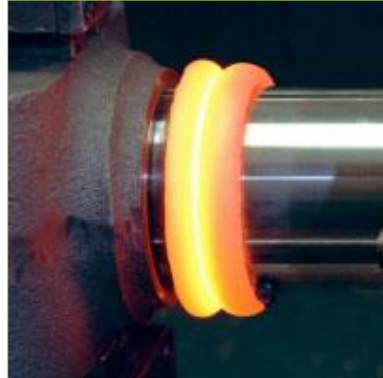
The angel of the North



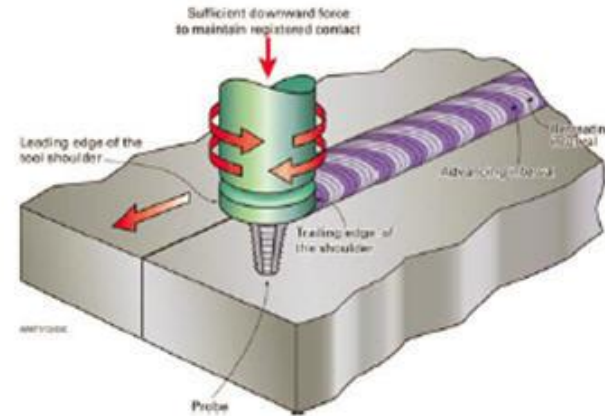
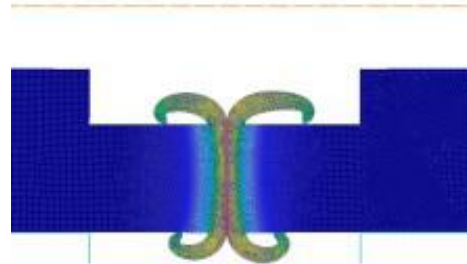
Offshore oil rig



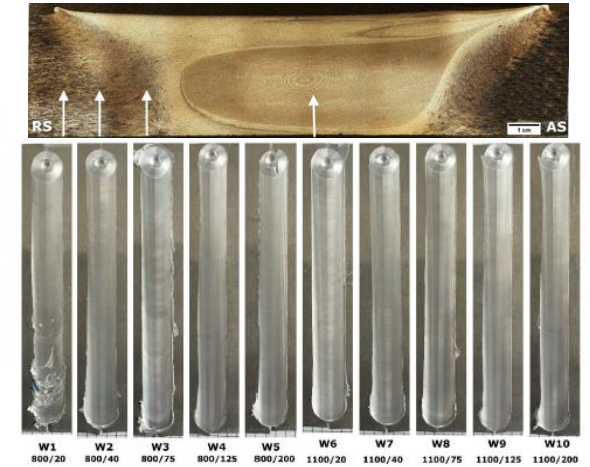
# Solid State Welding Processes



**Inertia friction welding for  
areoengine shaft**



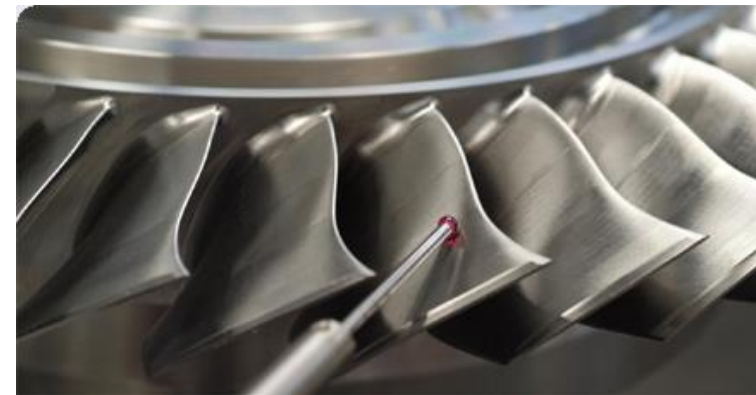
**Friction stir welding (FSW),  
Invention by TWI**  
<https://www.youtube.com/watch?v=y7rCTdxvGlg>



**O. Al-Jumaili, et al. FSW of Al6082,  
JMPT, vol.275,2020**

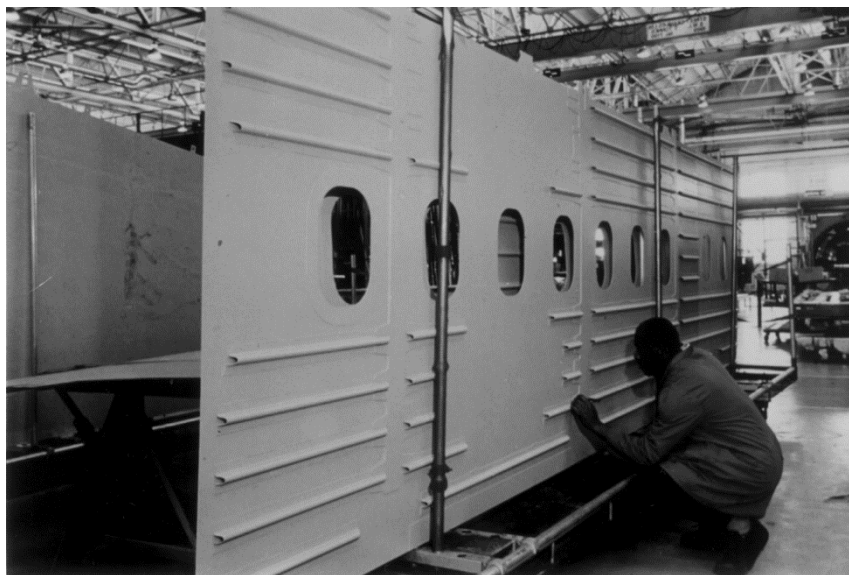


**Inertia friction welding machine, MTC**



**Linear friction welding for  
manufacturing of blisk, MTU**

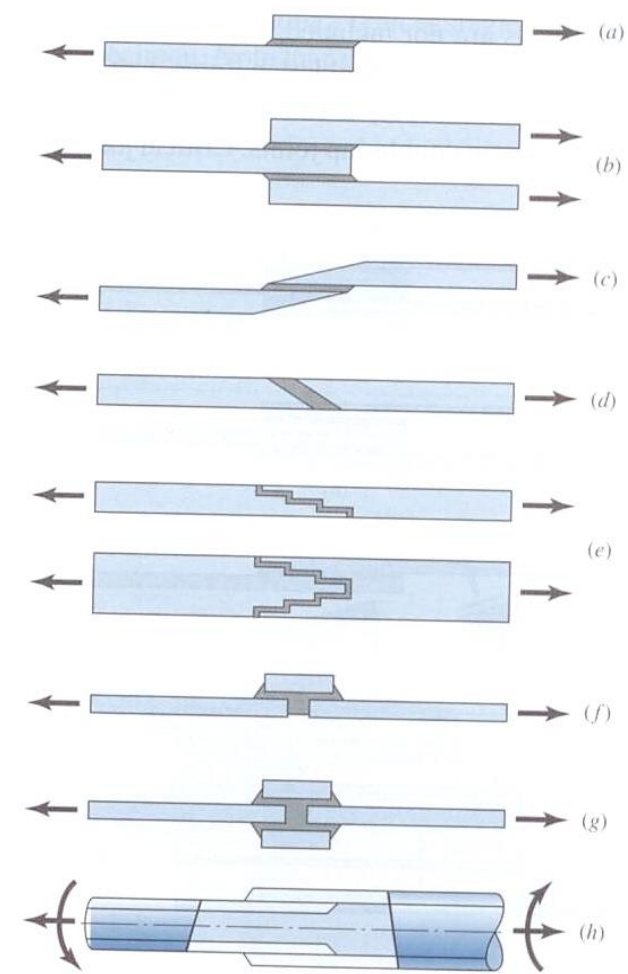
# Adhesive joints



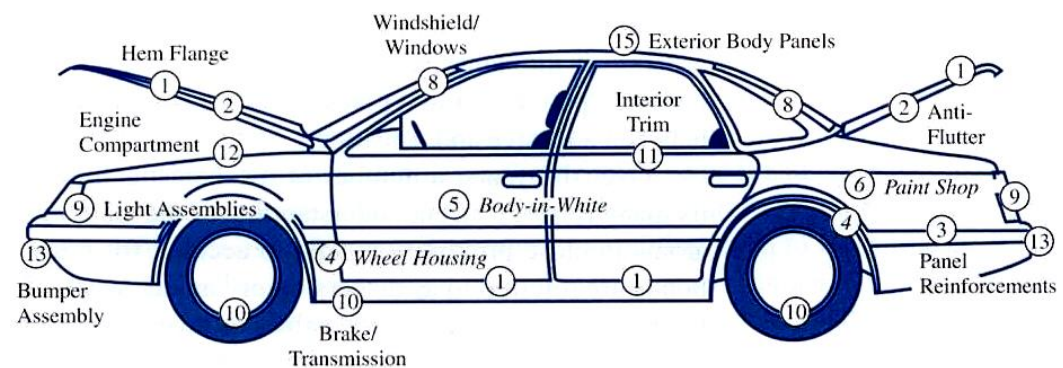
**BAe RJ fuselage panel with stringers bonded to skin**



**Adhesive joint of engine component**



**Types of adhesive lap joints**



**15 locations of adhesives in a car**

# Quiz 1: True or False to each of the following statements

## **bolted joints**

- A. **Bolted joint** may be used as both “permanent” and “non-permanent” joints.
- B. **Bolted joint** can be used to take both tensile and shear load.
- C. **Rivets** are good in taking tensile but not shear load.
- D. Welding is commonly used to provide “permanent” joint.

# Quiz 1: True or False to each of the following statements

## **bolted joints**

- A. **Bolted joint** may be used as both “permanent” and “non-permanent” joints. (true)
- B. **Bolted joint** can be used to take both tensile and shear load. (true)
- C. **Rivets** are good in taking tensile but not shear load. (false)
- D. Welding is commonly used to provide “permanent” joint. (true)



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# Design and analysis of bolted joints

## End of Part 1





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# Design and analysis of bolted joints

## Part 2

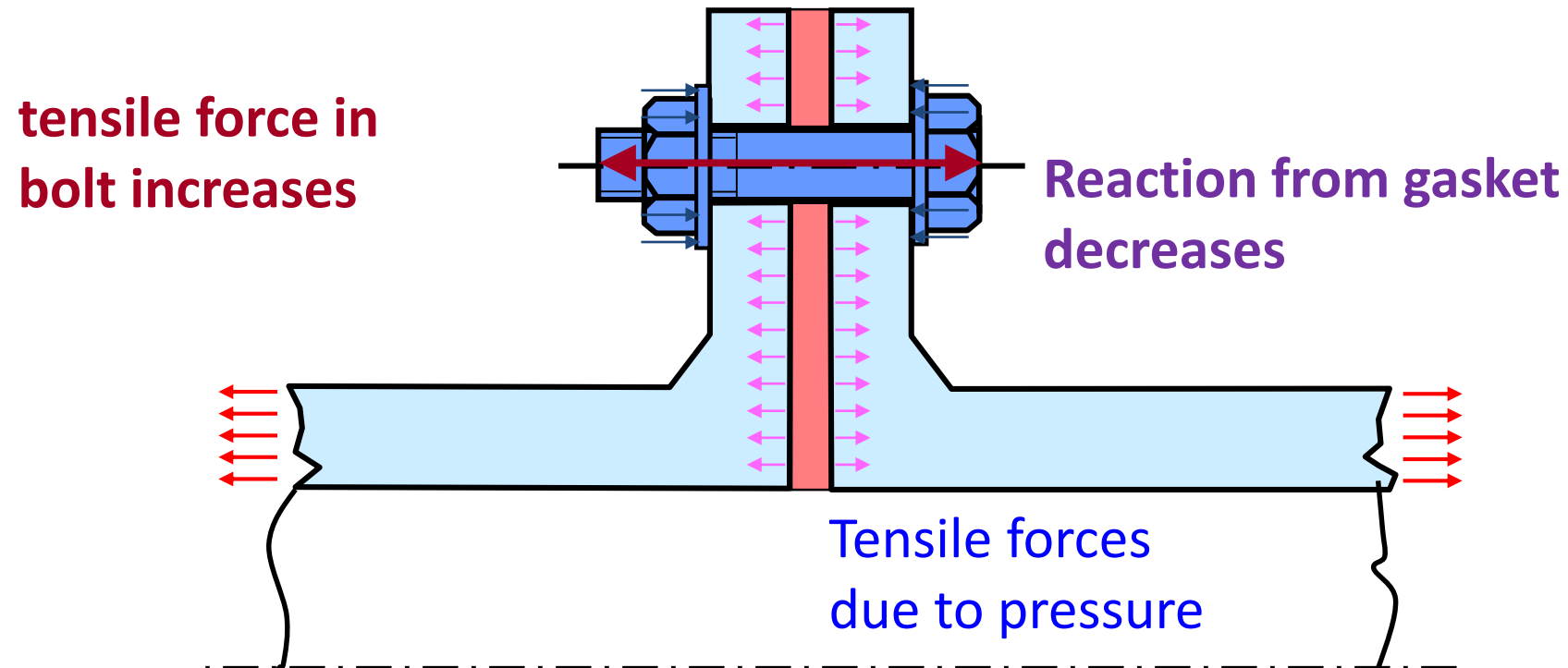
# Design Consideration of Bolted Joints

- Permanent or non-permanent joints
- Loads, stresses & strength
- Life
- Operational conditions
- Tooling & manufacturing efficiency



# Loading of a bolted joint

- When a load is applied to the joint (pressure, inertia, etc.), some of this load will stretch the bolt above its initial (pre-tensioned) length.



# Pre-tensioned bolted joints

Bolts are pre-tensioned for two reasons

1. The bolt force must exceed the maximum force in the clamped members in service otherwise the joint faces **will separate**.
2. Pre-tensioning reduces the fluctuating stresses experienced by the bolt, thereby increasing **the fatigue life**.



# Pre-tensioned bolted Joints

- Recommended pre-load for **non-permanent** joints

$$F_i = 0.75A_S\sigma_p$$

- Recommended pre-load for **permanent** joints

$$F_i = 0.9A_S\sigma_p$$

where,  $A_S$  is the **tensile area** of the bolt,

$\sigma_p$  is the **proof strength** of the bolt.

$$A_S = \frac{\pi}{16} (d_p + d_r)^2 \quad \begin{array}{ll} d_p = d - 0.6495p & \text{(pitch diameter)} \\ d_r = d - 1.0825p & \text{(minor diameter)} \end{array}$$

( $d$  is nominal major diameter and  $p$  is pitch)

If detailed information of the **proof strength**,  $\sigma_p$  is unavailable, an approximate value may be used

$$\sigma_p = 0.85\sigma_y \quad \sigma_y \text{ is the } \mathbf{yield\ strength}$$

# Grades & markings of bolts

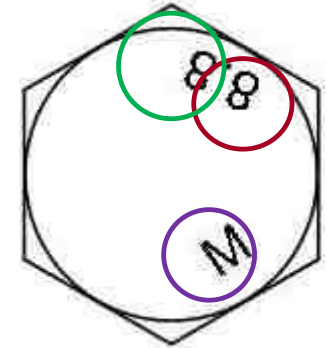
- **BS 3692:2001 ISO metric hexagon bolts, screws and nuts – Specification**

**Notes:** ISO stands for International Organisation of Standards

M – ISO metric thread

1<sup>st</sup> 8 represents **100<sup>th</sup>** of the **tensile Strength**

2<sup>nd</sup> 8 or 0.8 indicates the **ratio** between the **yield strength** and the **tensile strength**



For example of an **8.8** bolt,

the **tensile strength**:  $\sigma_{UTS} = 8 \times 100 = 800(MPa)$

the **yield strength**  $\sigma_y = 0.8 \times 800 = 640(MPa)$

Therefore,  $\sigma_p = 0.85\sigma_y = 0.85 \times 640 = 544(MPa)$

# Metric thread tensile area and mechanical properties

[Bolted joint Part 2](#)

- **BS EN ISO 898-1: 2013** Mechanical properties of fasteners made of carbon and alloy steels (page 11)

Note: **EN** denotes European standards

Table 5 — Proof loads — ISO metric coarse pitch thread

Thread <sup>a</sup> <i>d</i>	Nominal stress area <i>A<sub>s,nom</sub></i> mm <sup>2</sup>	Property class								
		4.6	4.8	5.6	5.8	6.8	8.8	9.8	10.9	12.9/12.9
Proof load, $F_p$ ( $A_{s,nom} \times S_{p,nom}$ ), N										
M3	5,03	1 130	1 560	1 410	1 910	2 210	2 920	3 270	4 180	4 880
M3,5	6,78	1 530	2 100	1 900	2 580	2 980	3 940	4 410	5 630	6 580
M4	8,78	1 980	2 720	2 460	3 340	3 860	5 100	5 710	7 290	8 520
M5	14,2	3 200	4 400	3 980	5 400	6 250	8 230	9 230	11 800	13 800
M6	20,1	4 520	6 230	5 630	7 640	8 840	11 600	13 100	16 700	19 500
M7	28,9	6 500	8 960	8 090	11 000	12 700	16 800	18 800	24 000	28 000
M8	36,6	8 240 <sup>c</sup>	11 400	10 200 <sup>c</sup>	13 900	16 100	21 200 <sup>c</sup>	23 800	30 400 <sup>c</sup>	35 500
M10	58	13 000 <sup>c</sup>	18 000	16 200 <sup>c</sup>	22 000	25 500	33 700 <sup>c</sup>	37 700	48 100 <sup>c</sup>	56 300
M12	84,3	19 000	26 100	23 600	32 000	37 100	48 900 <sup>d</sup>	54 800	70 000	81 800
M14	115	25 900	35 600	32 200	43 700	50 600	66 700 <sup>d</sup>	74 800	95 500	112 000
M16	157	35 300	48 700	44 000	59 700	69 100	91 000 <sup>d</sup>	102 000	130 000	152 000
M18	192	43 200	59 500	53 800	73 000	84 500	115 000	—	159 000	186 000
M20	245	55 100	76 000	68 600	93 100	108 000	147 000	—	203 000	238 000
M22	303	68 200	93 900	84 800	115 000	133 000	182 000	—	252 000	294 000
M24	353	79 400	109 000	98 800	134 000	155 000	212 000	—	293 000	342 000

Thread nominal dia, *d*

Tensile area, *A<sub>s</sub>* (mm<sup>2</sup>)

Thread property grade, (X.Y)

Proof load  $F_p = A_s \times \sigma_p$  (N)

# Determining bolt torque to pre-tension

- Recall torque equation for power screw

$$T = \frac{F_i d_p (\mu \pi d_p + l \cos \alpha)}{2(\pi d_p \cos \alpha - \mu l)} + \mu_c F_i \frac{d_c}{2}$$

where  $d_p$  is pitch diameter,  $d_c$  is collar diameter,  $l$  is lead,  $F_i$  is pre-tension,  $\mu$  and  $\mu_c$  are friction coefficients.

rearrange and simplify the above equation

$$T = K F_i d$$

where  $K$  is torque coefficient dependent upon surface finish & lubrication,  $K \approx 0.2$  for most cases;  $d$  is nominal diameter and  $F_i$  is pre-tension



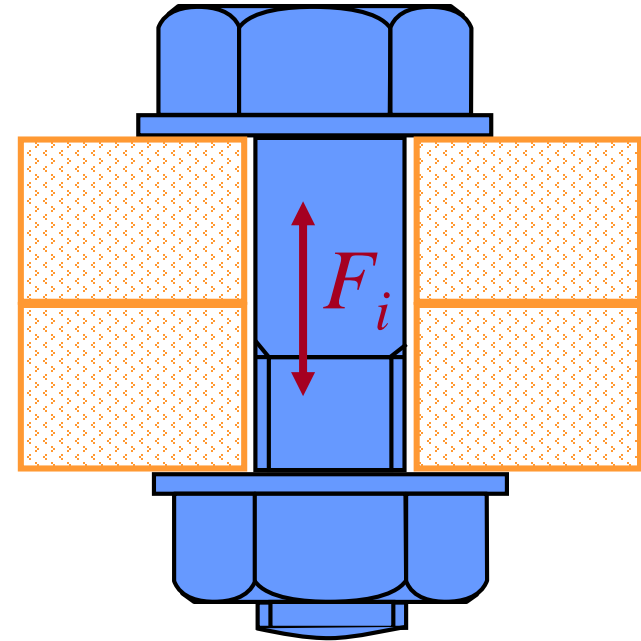
# Modelling of bolted joint

$F_i$  = preload on bolt due to tightening

$F_b$  = resultant load on bolt in tension

$F_c$  = resultant load on components in compression

$$F_i = F_b = F_c$$



# Modelling of bolted joint

$F_i$  = preload on bolt due to tightening

$P$  = external tensile load:

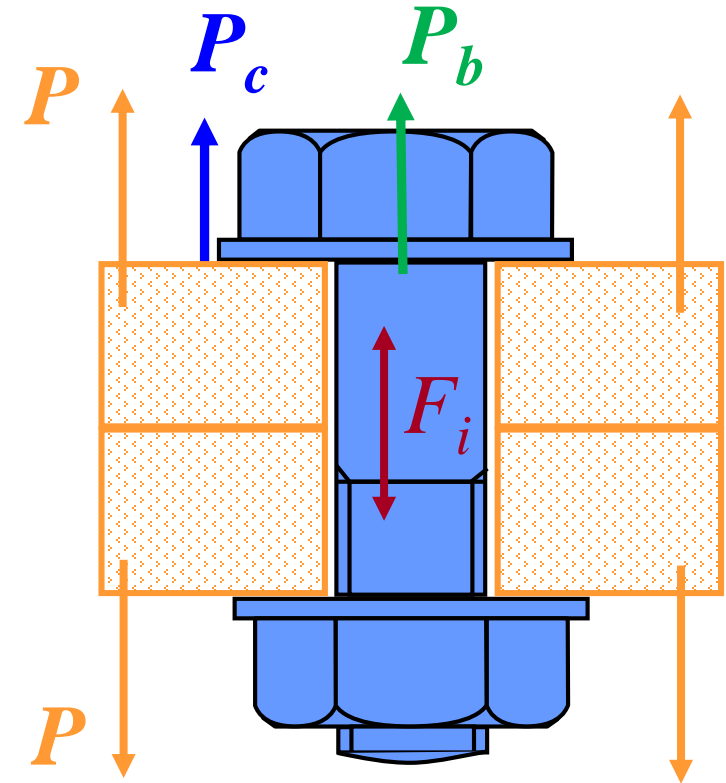
$$P = P_c + P_b$$

$P_b$  = portion of  $P$  taken by bolt

$P_c$  = portion of  $P$  taken by components

$F_b$  = resultant load on bolt

$F_c$  = resultant load on components

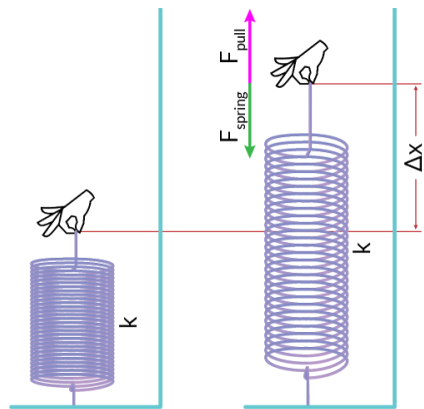


$$F_b = P_b + F_i > 0$$

$$F_c = P_c - F_i \leq 0$$



# Modelling of bolted joint



Hooke's law of a spring

$$F = k \times \Delta x$$

- Effectively, the bolted joint can be modelled as **two springs in parallel** with the components in compression & the bolt in tension.

Change in bolt length

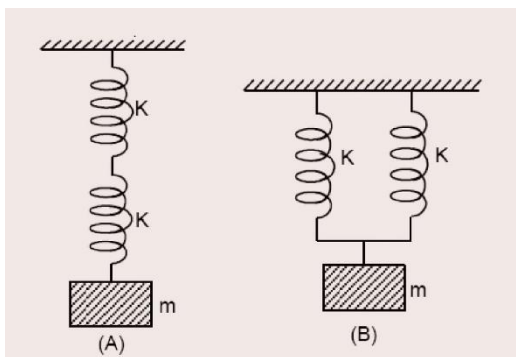
$$\delta_b = \frac{P_b}{K_b}$$

=

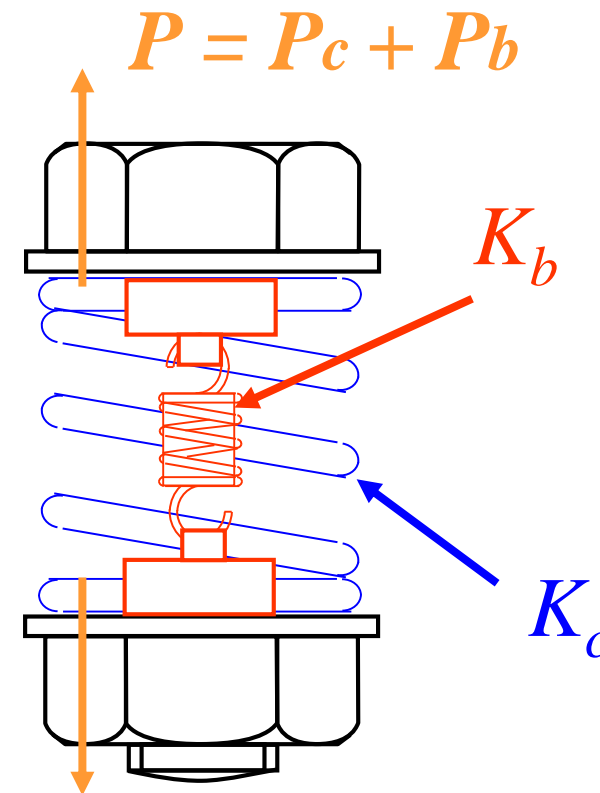
Change in components length

$$\delta_c = \frac{P_c}{K_c}$$

$$\delta_b = \delta_c$$



Springs in parallel & series



# Modelling of bolted joint

as  $\delta_b = \frac{P_b}{K_b} = \delta_c = \frac{P_c}{K_c}$  , then  $\frac{P_b}{K_b} = \frac{P_c}{K_c}$

but total external load is:

$$P = P_b + P_c = P_b + P_b \left( \frac{K_c}{K_b} \right) = \left( 1 + \frac{K_c}{K_b} \right) P_b$$

rearranging:

$$P_b = \frac{K_b}{K_b + K_c} P; \quad P_c = \frac{K_c}{K_b + K_c} P$$

# Modelling of bolted joint

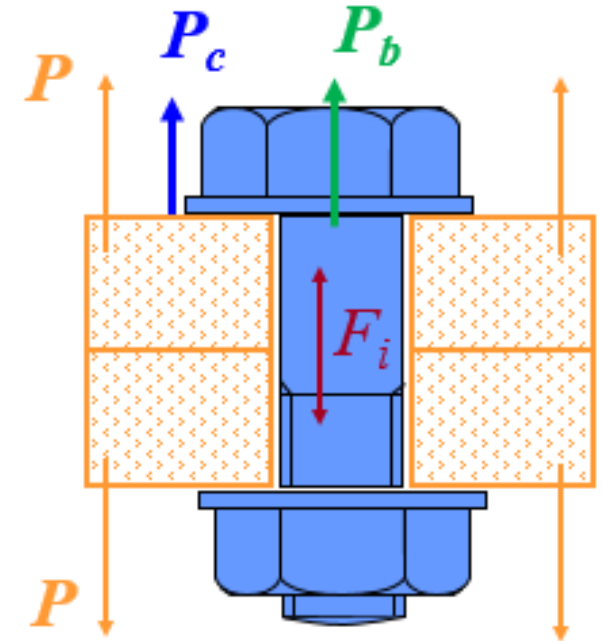
$$P_b = \frac{K_b}{K_b + K_c} P; \quad P_c = \frac{K_c}{K_b + K_c} P$$

Resultant load on the bolt **in tension** is:

$$F_b = P_b + F_i = \frac{K_b}{K_b + K_c} P + F_i > 0$$

Similarly, for the component **in compression**:

$$F_c = P_c - F_i = \frac{K_c}{K_b + K_c} P - F_i \leq 0$$



$$F_b = P_b + F_i > 0$$

$$F_c = P_c - F_i \leq 0$$

# Calculating $K_b$

- $K_b$  is easy enough to calculate as  $K_b = P/\delta$  if the bolt is threaded along its **length of grip**:

$$K_b = \frac{A_s E}{l_t}$$

where,  $A_s$  is tensile area,  $E$  is the Young's Modulus and  $l_t$  is the length

$$A_s = \frac{\pi}{16} (d_p + d_r)^2$$

$$d_p = d - 0.6495p$$

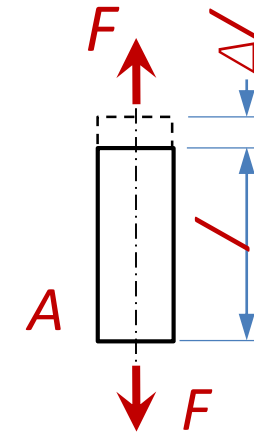
$$d_r = d - 1.0825p$$

(for metric threads)

where  $d_p$  is pitch diameter,  $d_r$  is minor diameter,  $d$  is nominal major diameter and  $p$  is pitch, respectively.

Alternatively,  $A_s$  may be found from **Table 5, BS EN ISO 898-1: 2013**

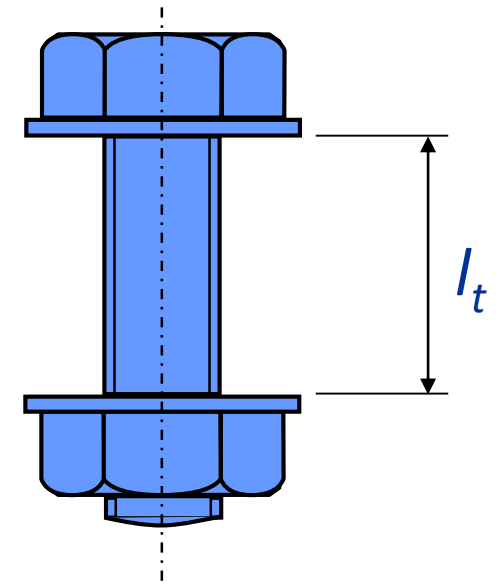
## Stress-strain relation



$$\sigma = E \varepsilon$$

$$\frac{F}{A} = E \frac{\Delta l}{l}$$

$$F = \frac{AE}{l} \Delta l$$

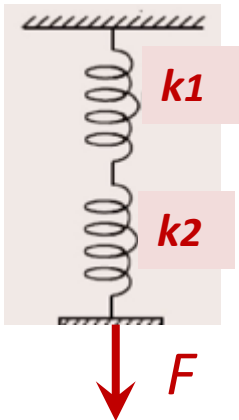


# Calculating $K_b$

- If there is an unthreaded portion of cross-sectional area  $A_d$  and length  $l_d$  (including necked bolts) then use formula for **springs in series** to give:

$$K_b = \frac{A_d A_s E}{A_d l_t + A_s l_d}$$

- How to derive the above formula?**



The derivation of the above equation is based on the concept of **Equivalent Stiffness of springs in series**

$$\frac{1}{k} = \frac{1}{k_1} + \frac{1}{k_2} + \dots$$

which may be derived by using the following relations

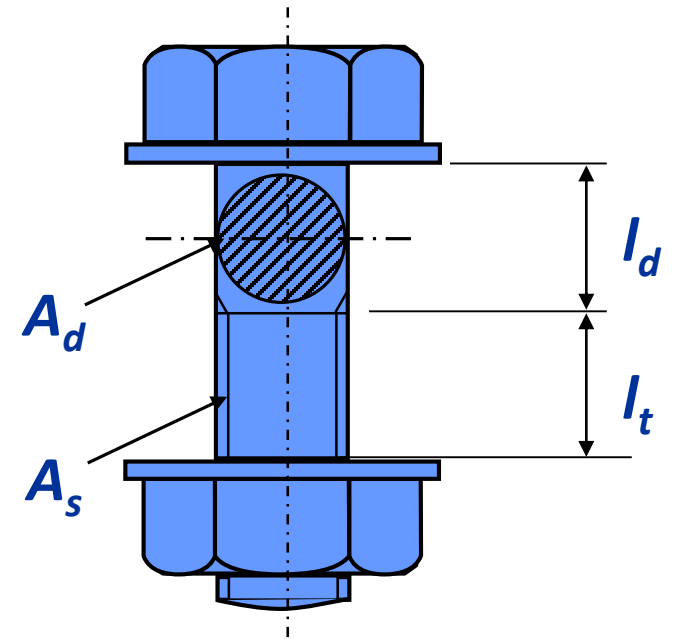
$$F = F_1 = F_2$$

$$F = k \Delta l$$

$$F_1 = \frac{A_1 E}{l_1} \Delta l_1$$

$$F_2 = \frac{A_2 E}{l_2} \Delta l_2$$

$$\Delta l = \Delta l_1 + \Delta l_2$$

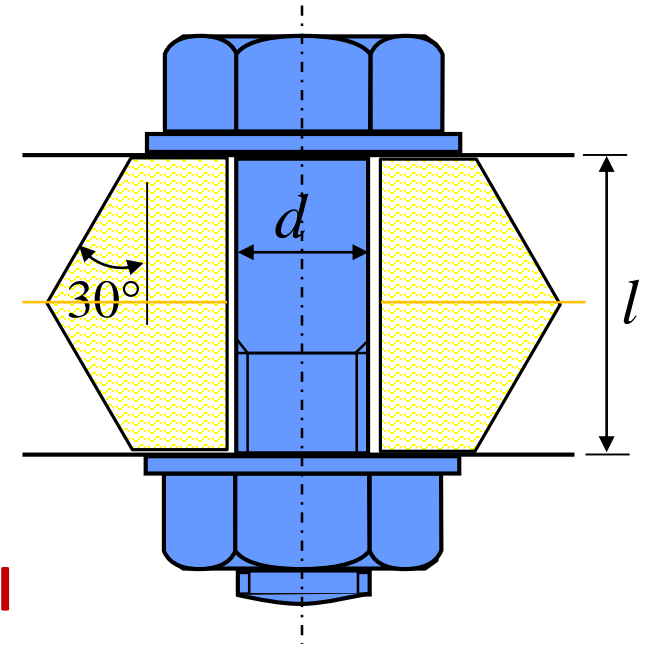


# Calculating $K_c$

- **$K_c$  is harder**: requires us to consider hollow cone-shaped regions in compression
- It can be shown (**no need to know how to derive**):

$$K_c = \frac{0.5774\pi E d}{2 \ln \left( 5 \frac{0.5774l + 0.5d}{0.5774l + 2.5d} \right)}$$

(detailed derivations are given in **Shigley et al., Mechanical Engineering Design, Ch8, TJ230 SHI** )





# Concept of “hard” joint

- A “good” joint would have **stiff components** and **elastic bolts**
- In a well engineered **‘hard’ joint**, the components are stiff  $K_c \gg K_b$ :
  - $K_c$  might be 1.7 GN/m or 2.2 GN/m
  - whereas  $K_b$  would be 250 MN/m or 500 MN/m
  - i.e.  **$K_c > 3 K_b$**
- In **‘soft’ joint**, the bolt is stiffer than the components which results in:
  - High level of bolt **fatigue** loads

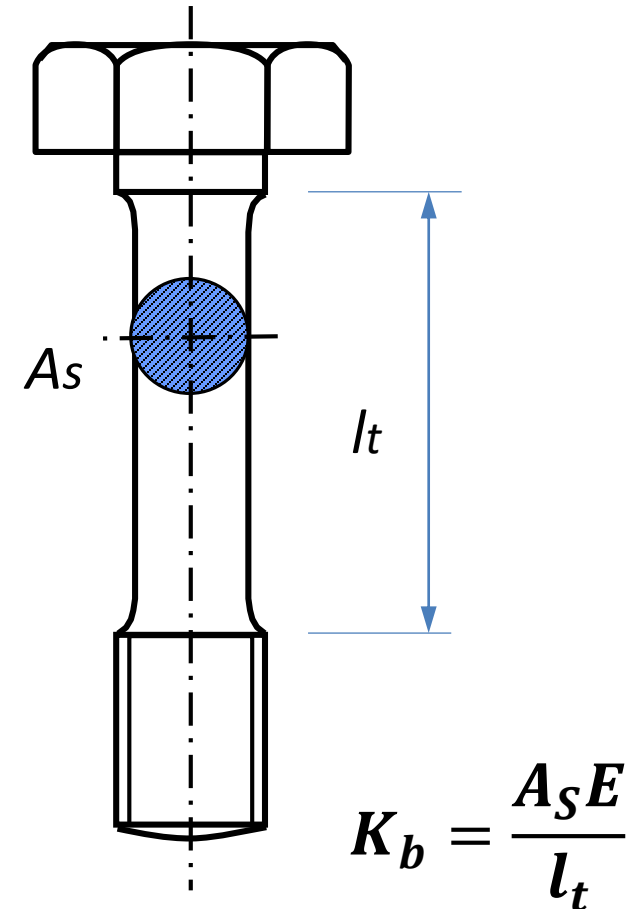
**Recall,**

$$F_b = \frac{K_b}{K_b + K_c} P + F_i > 0$$

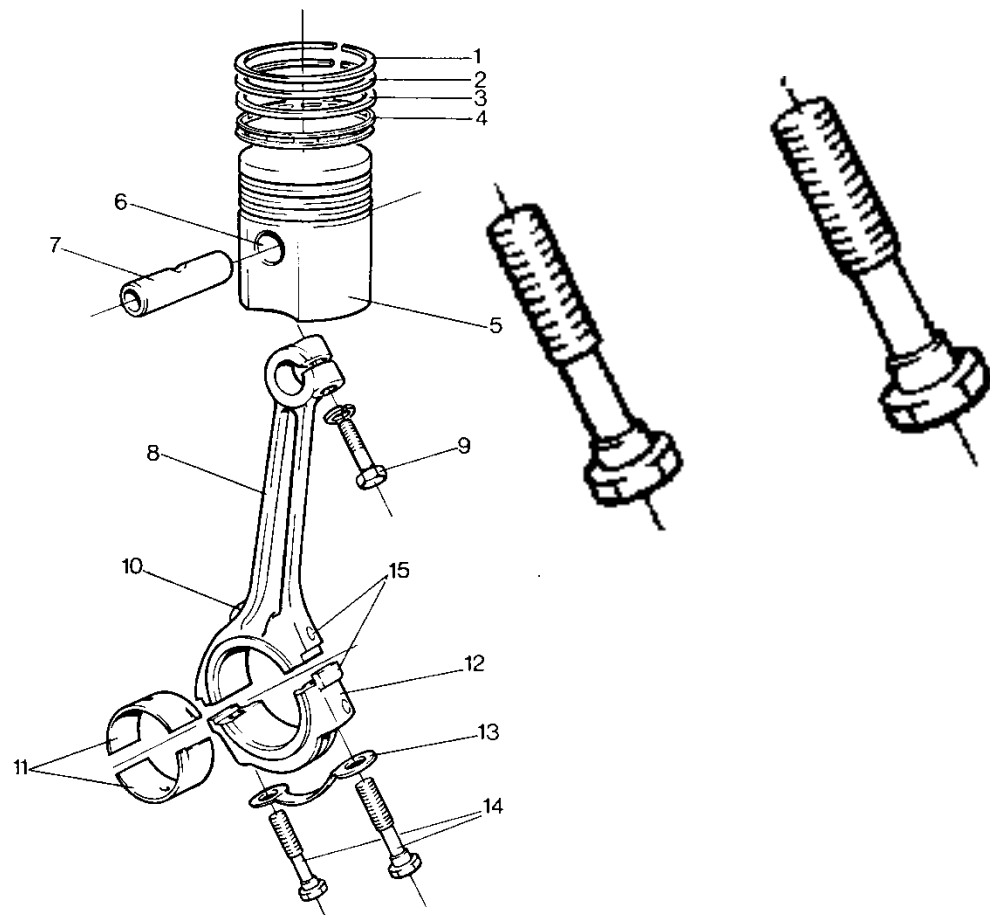
$$F_c = \frac{K_c}{K_b + K_c} P - F_i \leq 0$$

# Reducing bolt stiffness: necked bolts

- For a bolt ( $E$  is dependent on material) we require
  - (1) A **small** cross-sectional **area**
  - (2) A **long** length
- A small cross-sectional area
  - Necked down to root diameter
  - **Reduces stiffness without loss of strength**



# Reducing bolt stiffness: necked bolts



**Necked bolts** in connecting rod

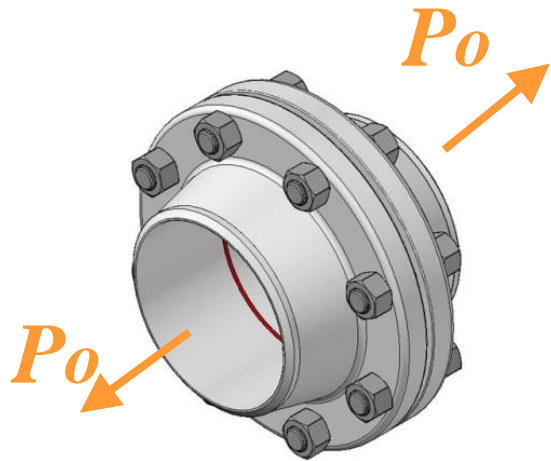


**Tie rods** used in pneumatic cylinders

**Reducing bolt stiffness:  
long bolts or tie rods to make a  
“hard” joint**

# Joint failure & reserve factor

- In application, normally **multiple (N) bolts** need to be used



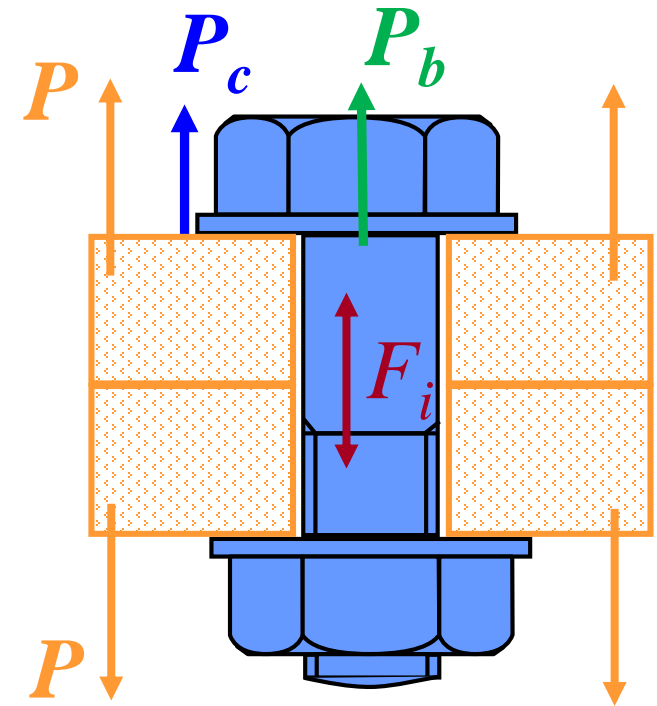
$$\frac{K_c}{K_b + K_c} P_0 - N F_i = 0$$

$$P_0 = N F_i \frac{K_b + K_c}{K_c}$$

Therefore, **reserve factor**  $n_0 = \frac{P_0}{P} \geq 1.5 \sim 2$

$P_0$  is the **maximum allowable external load** applied to **N bolts** at  $F_i$  preload,


$P$  is the **actual external load** applied to the bolted joint



$$F_b = \frac{K_b}{K_b + K_c} P + F_i > 0$$

$$F_c = \frac{K_c}{K_b + K_c} P - F_i \leq 0$$

# A general guide for reserve factor

Recommended reserve factor ( $n_o$ )	Operational and environmental conditions as well as use of materials	Uncertainty of material and working conditions
<b>1.25 ~ 1.5</b>	Reliable materials under controlled conditions subjected to loads and stresses known with certainty	
<b>1.5 ~ 2</b>	Well-known materials under reasonably constant environmental conditions subjected to known loads and stresses	
<b>2 ~ 2.5</b>	Average materials subjected to known loads and stresses	
<b>2.5 ~ 3.0</b>	Less well-known materials under average conditions of load, stress, and environment	
<b>3 ~ 3.4</b>	Untried materials under average conditions of load, stress, and environment or well-known materials under uncertain conditions of load, stress, and environment	

# Design of pre-tensioned bolted joints

1. Consider to use a non-permanent or permanent joint, define external load ( $P$ ) and reserve factor ( $no$ ), decide the number of bolts ( $N$ ) to ensure no joint separation
2. Estimate preload ( $F_i^{est}$ ) of bolt by assuming a hard joint, e.g.  $K_c \approx 3K_b$ ,  $NF_i \geq \frac{K_c}{K_b + K_c} P$
3. Choose a suitable bolt size, e.g. M4, M6 or M10, ... and grade, e.g. 6.8, 8.8 or 10.9 and determine the preload ( $F_i$ ) by calculating or using **Table 5 of BS EN ISO 898-1: 2009**

$$F_i = 0.75A_s\sigma_p \text{ (non-permanent) or } F_i = 0.9A_s\sigma_p \text{ (permanent joint)} \geq F_i^{est}$$

4. Calculate the stiffness of the bolts & components ( $K_b$  &  $K_c$ ):

$$K_b = \frac{A_d A_s E}{A_d l_t + A_s l_d} \quad \text{or} \quad K_b = \frac{A_s E}{l_t} \quad \text{and} \quad K_c = \frac{0.5774\pi E d}{2 \ln \left( 5 \frac{0.5774l + 0.5d}{0.5774l + 2.5d} \right)}$$

6. Calculate the maximum allowable external load ( $P_0$ ):

$$P_0 = NF_i \frac{K_b + K_c}{K_c}$$

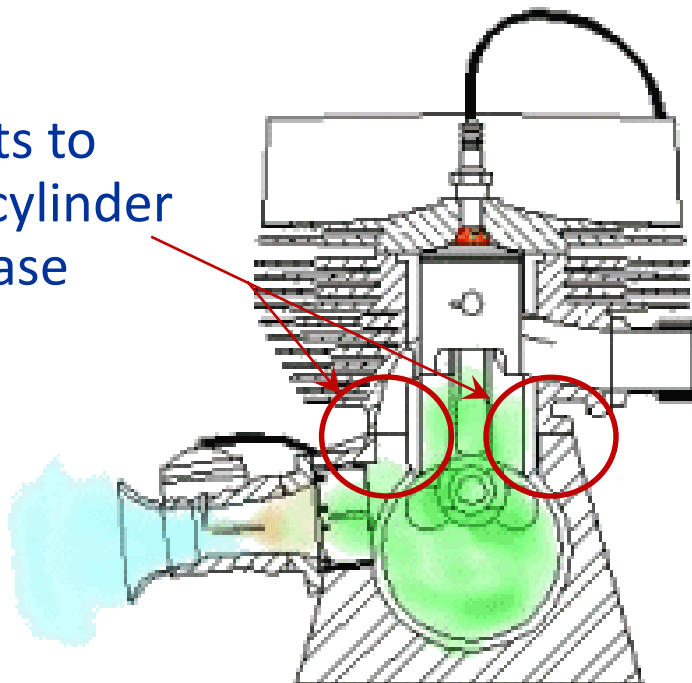
7. Calculate the reserve factor,  $no = \frac{P_0}{P} \geq 1.5 \sim 2$ , if NOT, go back to Step 3 & iterate



# Bolted joint under cyclic loading

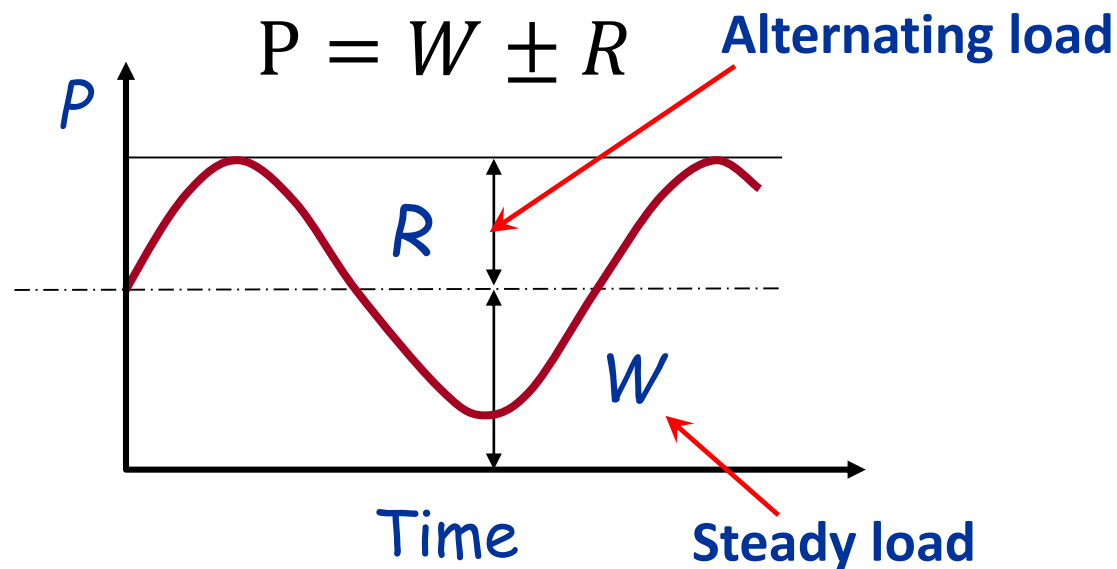
- Advantages of joints under cyclic loading, e.g. the **bolted joint of the cylinder head & crank case of a 2-stroke engine**:

Bolted joints to clamp the cylinder and crankcase



Example of a small two-stroke engine

[https://en.wikipedia.org/wiki/Two-stroke\\_engine](https://en.wikipedia.org/wiki/Two-stroke_engine)



$$F_c = \frac{K_c}{K_b + K_c} P - F_i \leq 0$$

$$F_i \geq (W + R) \frac{K_c}{K_b + K_c}$$

- For cylinder and crankcase faces to remain in contact:

# Bolted joint under cyclic loading

Resultant force in bolt:

We know,  $P = W \pm R$

$$F_b = (W \pm R) \frac{K_b}{K_b + K_c} + (W + R) \frac{K_c}{K_b + K_c}$$

$$F_i \geq (W + R) \frac{K_c}{K_b + K_c}$$

$$= \left( W + \frac{K_c}{K_b + K_c} R \right) \pm \frac{K_b}{K_b + K_c} R$$

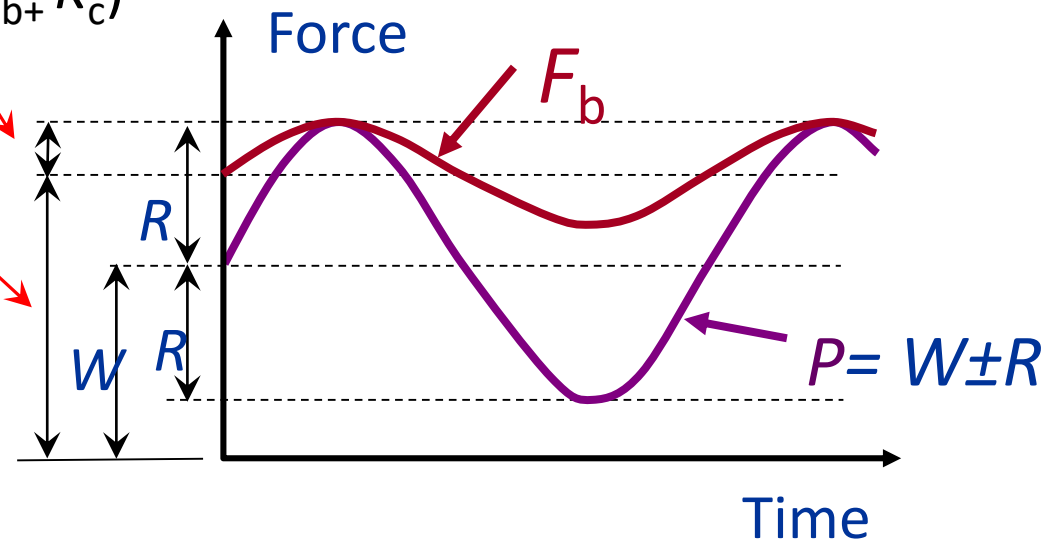
$$F_b = \frac{K_b}{K_b + K_c} P + F_i > 0$$

Steady load increased

Fluctuating load reduced

$R K_b / (K_b + K_c)$

$W + R K_c / (K_b + K_c)$



# Summary

- Be familiar with various joining techniques and different types of joints;
- Be able to calculate pre-load for **non-permanent and permanent bolted joints** based on **BS/ISO bolt strength grading**, e.g. M10 8.8;
- Be able to determine the **stiffness** of the bolt & clamped members, **resultant loads**;
- Understand the behaviour of pre-tensioned joint under **cyclic loading**;
- Be able to use **reserve factor formula** and the **suggested steps** to design a pre-tensioned bolt joint.

# Revision questions

- How to determine the recommended pre-tension load ( $F_i$ )?
- How much tightening torque ( $T$ ) is needed to achieve the recommended pre-tension load ( $F_i$ )?
- By picking up any metric bolts/screws to BS 3692 or BS/EN/ISO 898, how do we know their mechanical properties, e.g.  $\sigma_{UTS}$  or  $\sigma_p$ ?
- How much external load ( $P$ ) will be taken by the bolt ( $P_b$ ) and the components ( $P_c$ )?
- In designing a bolted joint, why is a “hard” joint a preferred option?
- Why is a pre-tensioned bolted joint is beneficial for cyclic loading?
- What is the threshold or critical requirement used in the design of a bolted joint?

# Bolted joint resources

- <http://www.tribology-abc.com>
  - Great calculators and other general information on threads (+ bearings)
- Childs, R.N., 2004. Mechanical Design, Elsevier
  - Chapter 12 discusses screw threads
- Shigley, J.E., Mischke, C.R., Budynas, R.G., 2003. Mechanical Engineering Design, 7<sup>th</sup> edition, McGraw-Hill, (TJ230 SHI)
  - Chapter 8 covers threads & joints



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# Design and analysis of bolted joints

End of Part 2



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# Design and analysis of bolted joints

## Part 3 Worked examples



# Worked example 1: Joint design of a 2-stroke engine

Bolted joint Part 3

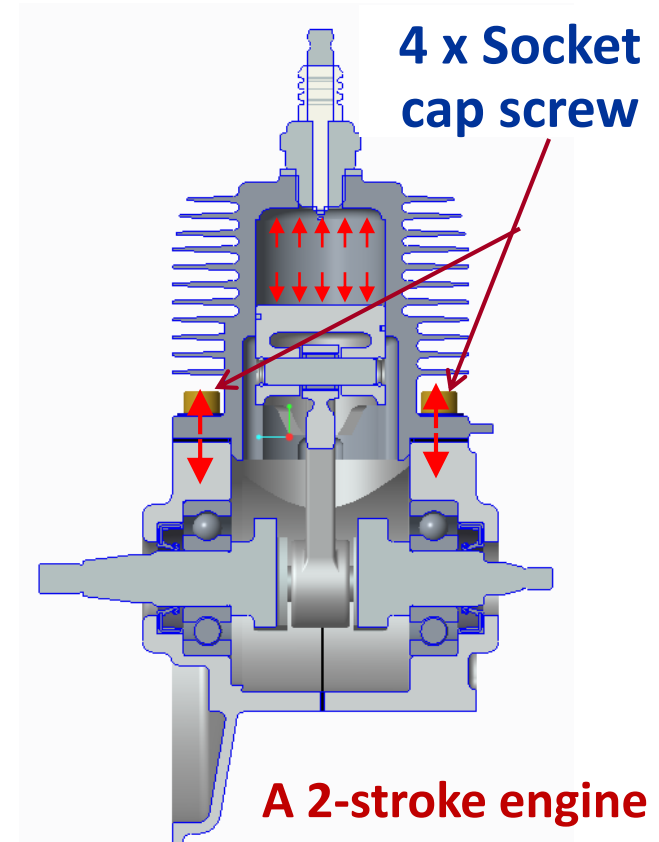
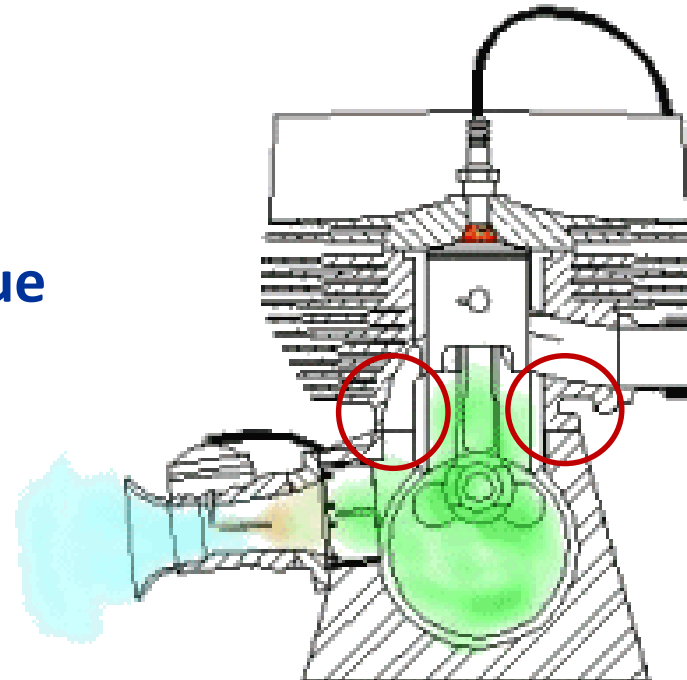
- Design bolted joint of a 2.4 kW 2-stroke engine.
  - Peak force is  $P = 6.5 \text{ kN}$
  - A **permanent joint** with a threaded grip length  $l_t = 25 \text{ mm}$
  - Cylinder and crankcase are made of cast Al ( $E=70 \text{ GPa}$ )
  - 4 x bolts (5.6 or similar, carbon steel,  $E=200 \text{ GPa}$ )
  - **Reserve factor** should be in the range of 1.5~2.0.

- Determine

- Suitable size of socket cap screw
- Right amount of tightening torque

Otto cycle of two-stroke engine

[https://en.wikipedia.org/wiki/Two-stroke\\_engine](https://en.wikipedia.org/wiki/Two-stroke_engine)



# Worked example 1: Joint selection of a 2-stroke engine

## a) Selection of a suitable socket cap screw

[Bolted joint Part 3](#)

1. Joint design specifies external load ( $P=6.5 \text{ kN}$ ) and reserve factor ( $n_o=1.5\sim 2$ ), a permanent joint of 4 x socket cap screw ( $N=4$ ).

2. Estimate pre-load ( $F_i$ ) of bolt by assuming a hard joint,  $K_c \approx 3K_b$ ,  $NF_i \geq \frac{K_c}{K_b+K_c} P$

$$F_i^{est} \geq \frac{1}{N} \frac{K_c}{K_b + K_c} P = \frac{1}{4} \times \frac{1}{1.3} \times 6,500 = 1,250(N)$$

3. Choose a suitable bolt size from **BS ISO 898-1: 2009 (Table 5)**

**M4** seems to be the right size

Table 5 — Proof loads — ISO metric coarse pitch thread

Thread <sup>a</sup> <i>d</i>	Nominal stress area <i>A<sub>s,nom</sub></i> <sup>b</sup> mm <sup>2</sup>	Property class						
		4.6	4.8	5.6	5.8	6.8	8.8	9.8
		Proof load, <i>F<sub>p</sub></i> ( <i>A<sub>s,nom</sub></i> × <i>S<sub>p,nom</sub></i> ), N						
M3	5,03	1 130	1 560	1 410	1 910	2 210	2 920	3 270
M3,5	6,78	1 530	2 100	1 980	2 580	2 980	3 940	4 410
M4	8,78	1 980	2 720	2 460	3 340	3 860	5 100	5 710
M5	14,2	3 200	4 400	3 980	5 400	6 250	8 230	9 230
M6	20,1	4 520	6 230	5 630	7 640	8 840	11 600	13 100
M7	28,9	6 500	8 960	8 090	11 000	12 700	16 800	18 800

$$F_i = 0.9 \times A_S \sigma_P = 0.9 \times F_P = 0.9 \times 2,460 = 2,214(N)$$

$$F_i = 2,214 > F_i^{est} = 1,250 (N)$$

Ok for detailed evaluation

# Worked example 1: Joint selection of a 2-stroke engine

Bolted joint Part 3

4. Calculate stiffness of the bolts & components ( $K_b$  &  $K_c$ ):

$$K_b = \frac{A_s E}{l_t} = \frac{8.78 \times 200 \times 10^3}{25} = 70.2 \times 10^3 (N/mm)$$

$$K_c = \frac{0.5774 \pi E d}{2 \ln \left( 5 \frac{0.5774 l + 0.5 d}{0.5774 l + 2.5 d} \right)} = \frac{0.5774 \times 3.1416 \times 70 \times 10^3 \times 4}{2 \times \ln \left( 5 \times \frac{0.5774 \times 25 + 0.5 \times 4}{0.5774 \times 25 + 2.5 \times 4} \right)}$$
$$= 211.3 \times 10^3 (N/mm)$$

5. Calculate the maximum allowable external load ( $P_o$ ):

$$P_o = N F_i \frac{K_b + K_c}{K_c} = 4 \times 2,214 \times \frac{(70.2 + 211.3) \times 10^3}{211.3 \times 10^3} = 11,790 (N)$$

6. Calculate the reserve factor ( $n_o$ ):

$$n_o = \frac{P_o}{P} = \frac{11,790}{6,500} = 1.8$$

which is in the range of 1.5~2.0. Therefore, 5.6 M4 socket cap screw is a suitable choice.

# Worked example 1: Joint selection of a 2-stroke engine

## b) Calculation of the required torque for tightening

Use simplified torque tightening equation

$$T = KF_i d$$

where  $K$  is torque coefficient,  $K \approx 0.2$  for most cases;  
 $d$  is nominal diameter and  $F_i$  is pre-tension



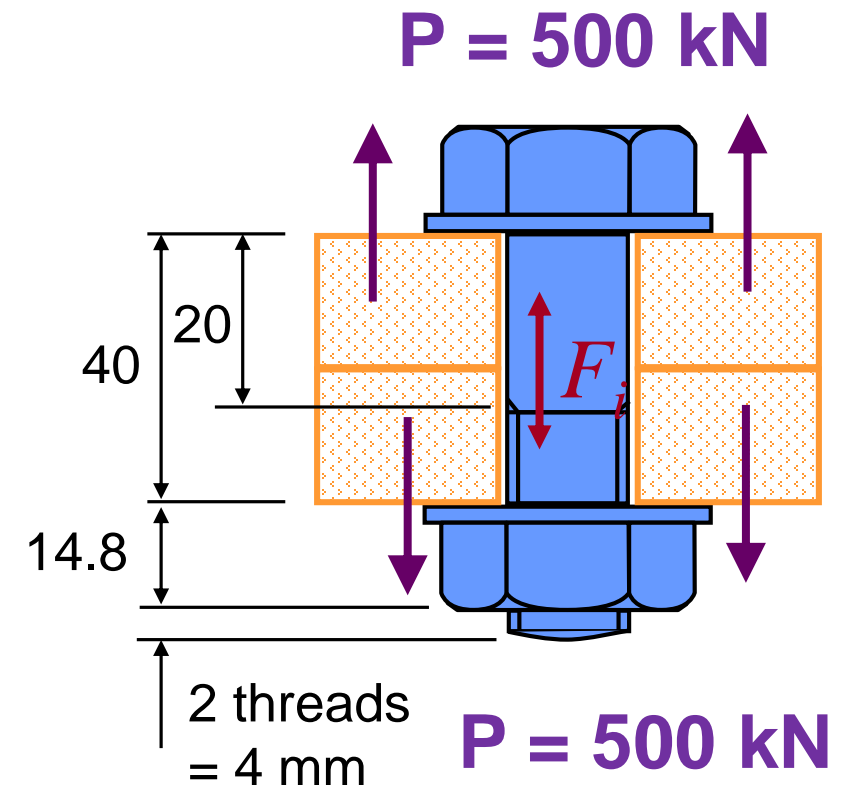
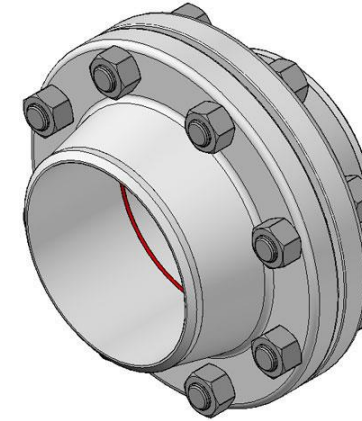
$$T = KF_i d = 0.2 \times 2,214 \times 4 \times 10^{-3} = 1.8 \text{ (Nm)}$$

Therefore, application of  $T=1.8 \text{ Nm}$  tightening torque to each **M4** socket cap screw would generate the required  $F_i=2,214 \text{ N}$  pre-tension load.

# Worked example 2: Evaluation of a bolted joint

## Evaluation of a designed bolt joint

- **Eight non-permanent steel M16, 8.8 bolts** ( $E = 200 \text{ GPa}$ ) are used to secure **40 mm** of non-gasketed cast iron flanged coupling ( $E = 96 \text{ GPa}$ ) on which is imposed a separating force of  **$P=500 \text{ kN}$** .
- **Determine:**
  - a) bolt stiffness  $K_b$
  - b) components stiffness  $K_c$
  - c) Reserve factor  $n_o$



# Bolt stiffness

- **Bolt stiffness:** 
$$K_b = \frac{A_d A_S E}{A_d l_t + A_S l_d}$$
- Area of unthreaded section 
$$A_d = \frac{\pi}{4} d^2 = \frac{\pi \times (16 \times 10^{-3})^2}{4} = 201.1 \times 10^{-6} \text{ m}^2$$
- Area of threaded section 
$$A_S = \frac{\pi}{16} (d_p + d_r)^2 = \frac{\pi}{16} (14.7 + 13.55)^2 \times 10^{-6}$$
  
$$= 156.7 \times 10^{-6} \text{ m}^2$$

$$K_b = \frac{201.1 \times 156.7 \times 10^{-12} \times 200 \times 10^9}{(201.1 \times 20 + 156.7 \times 20) \times 10^{-9}} = 881.6 \times 10^6 \text{ N/m}$$

- **Component stiffness:**

$$K_c = \frac{0.5774 \pi E d}{2 \ln \left( 5 \frac{0.5774 l + 0.5 d}{0.5774 l + 2.5 d} \right)} = 1,513.4 \times 10^6 \text{ N/m}$$

# Initial clamping force

- The recommended initial clamping force for **M16 8.8 non-permanent joints** is:

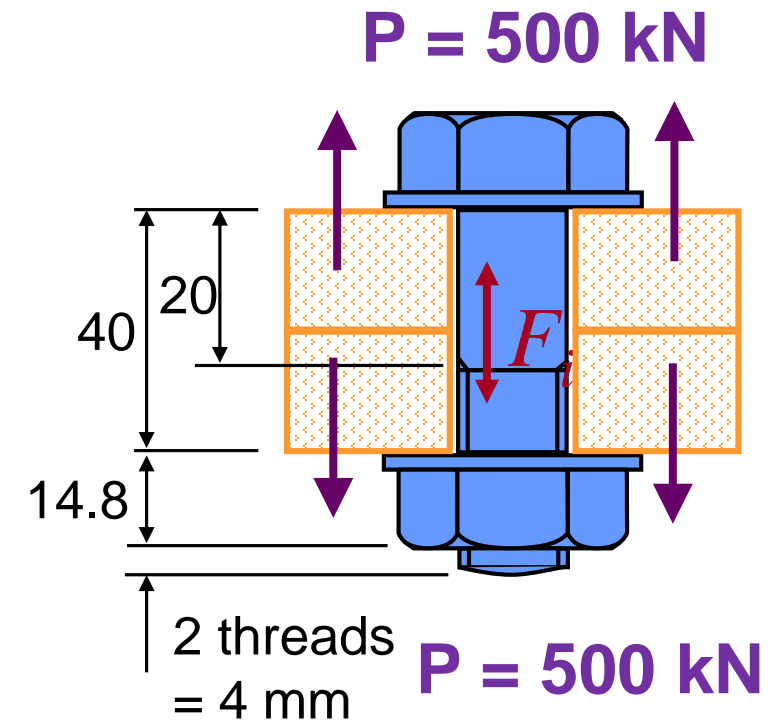
$$F_i = 0.75 A_S \sigma_p$$

$$\sigma_p = 0.85 \sigma_y = 0.85 \times 640 = 544 (MPa)$$

$$A_S = \frac{\pi}{16} (d_p + d_r)^2 = 156.7 \times 10^{-6} m^2$$

Therefore,

$$F_i = 0.75 \times 544 \times 156.7 = 63.9 \times 10^3 N$$





# Reserve factor

- Maximum external load to separation for **8 x M16 bolts**:

$$\begin{aligned}
 P_0 &= NF_i \frac{K_b + K_c}{K_c} \\
 &= 8 \times 63.9 \times 10^3 \frac{(881.6 + 1,513.4) \times 10^6}{1,513.4 \times 10^6} \\
 &= 809.2 \times 10^3 \text{ N}
 \end{aligned}$$

- Therefore, the **reserve factor** of this bolted joint is:

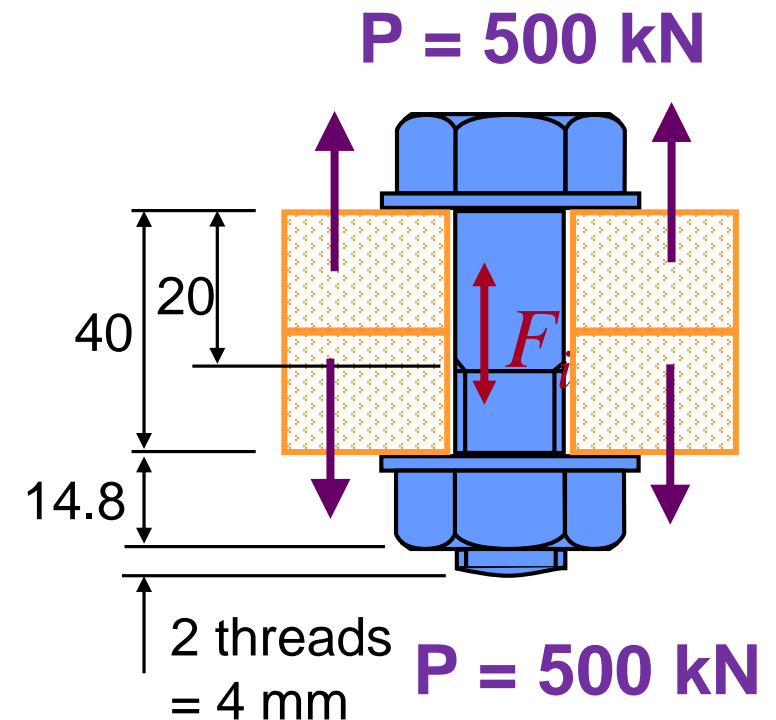
$$n_0 = \frac{P_0}{P} = \frac{809.2 \times 10^3}{500 \times 10^3} = 1.62$$

This is acceptable.

Recall,

$$\frac{K_c}{K_b + K_c} P_0 - NF_i = 0$$

$$P_0 = NF_i \frac{K_b + K_c}{K_c}$$



# Answers to revision questions

- The recommended **pre-tension load ( $F_i$ )**: (slide 24)

$$F_i = 0.75A_s\sigma_p \text{ (non-permanent) or } F_i = 0.9A_s\sigma_p \text{ (permanent joint)}$$

- Tightening torque ( $T$ )**: (slide 27)  $T = KF_i d$

- Mechanical properties** of metric bolts are defined by marks, e.g. **10.9** (slide 24)

- External load ( $P$ ) will be shared by the bolt ( $P_b$ ) and components ( $P_c$ ) (slides 29-32)

$$F_b = \frac{K_b}{K_b + K_c} P + F_i > 0 \qquad F_c = \frac{K_c}{K_b + K_c} P - F_i \leq 0$$

- A “**hard**” joint is preferred because components take a **larger load** (slides 36-38)

- Pre-tensioned bolted joint reduces **cyclic loading in bolts** (slides 42-43)

- The requirement** is to ensure **no joint separation** (slide 40)

$$P_0 = NF_i \frac{K_b + K_c}{K_c}$$

$$n_0 = \frac{P_0}{P} \geq 1.5 \sim 2$$



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# Design and analysis of bolted joints

**End of Session**